(SBA) guidelines. The Regulatory Flexibility Act requires that the Agency specifically consider the economic effects associated with its rules on small entities. The SBA defines a firm engaged in agriculture as "small" if it has less than \$750,000 in annual receipts.

Citrus producers in the area that is being removed from the list of quarantined areas will have greater choice of where to market their fruit. This should benefit producers by providing them with more alternatives. It is unlikely, however, that producer income or expenses will be affected in a measurable way.

It is difficult to quantify the benefits of removing an area from quarantine. While producers will have greater choice of where to market their citrus crops, most of the trees in the quarantined area have been destroyed. It is unlikely that a reduction in the quarantined area will have any measurable effect on producers or consumers.

Under these circumstances, the Administrator of the Animal and Plant Health Inspection Service has determined that this action will not have a significant economic impact on a substantial number of small entities.

Executive Order 12372

This program/activity is listed in the Catalog of Federal Domestic Assistance under No. 10.025 and is subject to Executive Order 12372, which requires intergovernmental consultation with State and local officials. (See 7 CFR part 3015, subpart V.)

Executive Order 12988

This final rule has been reviewed under Executive Order 12988, Civil Justice Reform. This rule: (1) Preempts all State and local laws and regulations that are inconsistent with this rule; (2) has no retroactive effect; and (3) does not require administrative proceedings before parties may file suit in court challenging this rule.

Paperwork Reduction Act

This final rule contains no new information collection or recordkeeping requirements under the Paperwork Reduction Act of 1995 (44 U.S.C. 3501 *et seq.*).

List of Subjects in 7 CFR Part 301

Agricultural commodities, Plant diseases and pests, Quarantine, Reporting and recordkeeping requirements, Transportation.

Accordingly, we are amending 7 CFR part 301 as follows:

PART 301—DOMESTIC QUARANTINE NOTICES

1. The authority citation for part 301 continues to read as follows:

Authority: 7 U.S.C. 166, 7711, 7712, 7714, 7731, 7735, 7751, 7752, 7753, and 7754; 7 CFR 2.22, 2.80, and 371.3.

Section 301.75–15 also issued under Sec. 204, Title II, Pub. L. 106–113, 113 Stat. 1501A–293; sections 301.75–15 and 301.75–16 also issued under Sec. 203, Title II, Pub. L. 106–224, 114 Stat. 400 (7 U.S.C. 1421 note).

2. In § 301.75–4, paragraph (a), in the entry for Manatee County, the second paragraph is revised to read as follows:

§ 301.75-4 Quarantined areas.

(a) * * *

Florida

* * * * *

Manatee County. * * *

That portion of the county bounded by a line drawn as follows: Beginning at the northwest corner of sec. 24, T. 33 S., R. 17 E.; then east along the northern boundary of sec. 24, T. 33. S., R. 17 E. (Bishop Harbor Road) until it becomes SR 683 (Moccasin Wallow Road); then east on SR 683 to the northeast boundary of sec. 22, T. 33 S., R. 18 E., then south along the eastern boundary of sec. 22, T. 33 S., R. 18 E. to 69th Street East; then east on 69th Street East to Erie Road: then south on Erie Road to U.S. Highway 301; then south on U.S. Highway 301 to Interstate 75; then south on Interstate 75 to the southern boundary of sec. 24, T. 35 S., R. 18 E.; then west along the southern boundaries of secs. 24, 23, and 22 to where the southern boundary of sec. 22 meets Whitfield Avenue; then west on Whitfield Avenue to U.S. Highway 301; then north on U.S. Highway 301 to SR 70; then west on SR 70 to U.S. Highway 41; then north on U.S. Highway 41 to where it becomes 14th Street West; then north on 14th Street West to 1st Avenue West: then east on 1st Avenue West to 9th Street West; then north on 9th Street West to the north bank of the Manatee River; then west along the north bank of the Manatee River to Terra Ceia Bay; then north along the western boundaries of secs. 25 and 24 to the point of the beginning.

Done in Washington, DC, this 25th day of February, 2002.

W. Ron DeHaven,

Acting Administrator, Animal and Plant Health Inspection Service.

[FR Doc. 02–4911 Filed 2–28–02; 8:45 am] BILLING CODE 3410–34-U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001-SW-62-AD; Amendment 39-12664; AD 2002-04-07]

RIN 2120-AA64

Airworthiness Directives; Eurocopter France Model AS350BA and B2 Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments.

SUMMARY: This amendment adopts a new airworthiness directive (AD) for Eurocopter France (ECF) Model AS350BA and B2 helicopters modified with a Eurocopter Canada Limited (ECL) Left-side-Pilot Configuration kit in accordance with Canadian Supplemental Type Certificate (STC) SH96-32 or United States STC SR00429 NY. This action requires replacing the collective locking device with a newlydesigned locking device. This amendment is prompted by a report of a locking device that engaged during flight. The actions specified in this AD are intended to prevent inadvertent engagement of a locking device, the collective pitch control locking in the full-down position, and subsequent loss of control of the helicopter.

DATES: Effective March 18, 2002.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of March 18, 2002

Comments for inclusion in the Rules Docket must be received on or before April 30, 2002.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Office of the Regional Counsel, Southwest Region, Attention: Rules Docket No. 2001–SW–62–AD, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137. You may also send comments electronically to the Rules Docket at the following address: 9-asw-adcomments@faa.gov.

The service information referenced in this AD may be obtained from Eurocopter Canada Limited, 1100 Gilmore Rd., Fort Erie, Ontario L2A 5M-4005, telephone (972) 641-3460, fax (972) 641-3527. This information may be examined at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Carroll Wright, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Regulations Group, Fort Worth, Texas 76193–0111, telephone (817) 222–5120, fax (817) 222–5961.

SUPPLEMENTARY INFORMATION: Transport Canada, the airworthiness authority for Canada, notified the FAA that an unsafe condition may exist on ECF Model AS350BA and B2 helicopters. Transport Canada advises of an incident involving an ECF Model AS350 helicopter, equipped with an ECL Left-Side Pilot Configuration kit, Canadian STC SH96–32 or United States STC SR00429 NY, in which the collective locking device engaged during flight.

ECL has issued Service Bulletin AS 350 BA, B2, No. ECL—99—67—002, Revision 2, dated September 23, 1999, which specifies replacing the collective locking device with a locking device that eliminates inadvertent engagement. Transport Canada classified this service bulletin as mandatory and issued AD No. CF—2000—06R1, dated August 23, 2000, to ensure the continued airworthiness of these helicopters in Canada.

These helicopter models are manufactured in France and are type certificated for operation in the United States under the provisions of 14 CFR 21.29 and the applicable bilateral agreement. The Left-Side Pilot Configuration kit was installed by ECL. Pursuant to the applicable bilateral agreement, Transport Canada has kept the FAA informed of the situation described above. The FAA has examined the findings of Transport Canada, reviewed all available information, and determined that AD action is necessary for products of these type designs that are certificated for operation in the United States.

This unsafe condition is likely to exist or develop on other helicopters of the same type design registered in the United States. Therefore, this AD is being issued to prevent inadvertent engagement of a collective locking device, the collective control locking in the full-down position, and subsequen loss of control of the helicopter. The actions must be accomplished in accordance with the service bulletin described previously. The short compliance time involved is required because the previously described critical unsafe condition can adversely affect the controllability of the helicopter. Therefore, replacing the locking device with a new locking device within 50 hours time-in-service or 30 days, whichever occurs first, is

required, and this AD must be issued immediately.

Since a situation exists that requires the immediate adoption of this regulation, it is found that notice and opportunity for prior public comment hereon are impracticable, and that good cause exists for making this amendment effective in less than 30 days.

The FAA estimates that 28 helicopters will be affected by this AD, that it will take approximately 5 work hours to install a locking device and that the average labor rate is \$60 per work hour. ECL has represented to the FAA that for each Left-Side Pilot Configuration delivered with a previous version of the collective locking device, one set of applicable parts specified in the service bulletin will be made available free of charge. Based on these figures, the total cost impact of the AD on U.S. operators is estimated to be \$8,400.

Comments Invited

Although this action is in the form of a final rule that involves requirements affecting flight safety and, thus, was not preceded by notice and an opportunity for public comment, comments are invited on this rule. Interested persons are invited to comment on this rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified under the caption ADDRESSES. All communications received on or before the closing date for comments will be considered, and this rule may be amended in light of the comments received. Factual information that supports the commenter's ideas and suggestions is extremely helpful in evaluating the effectiveness of the AD action and determining whether additional rulemaking action would be needed.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify the rule. All comments submitted will be available in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this AD will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their mailed comments submitted in response to this rule must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. 2001–SW–62–AD." The postcard will be date

stamped and returned to the commenter.

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

The FAA has determined that this regulation is an emergency regulation that must be issued immediately to correct an unsafe condition in aircraft. and that it is not a "significant regulatory action" under Executive Order 12866. It has been determined further that this action involves an emergency regulation under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979). If it is determined that this emergency regulation otherwise would be significant under DOT Regulatory Policies and Procedures, a final regulatory evaluation will be prepared and placed in the Rules Docket. A copy of it, if filed, may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive to read as follows:

2002–04–07 Eurocopter France: Amendment 39–12664. Docket No. 2001–SW–62–AD.

Applicability: Model AS350BA and B2 helicopters, certificated in any category, modified with a Eurocopter Canada Limited (ECL) Left-Side Pilot Configuration kit in accordance with Canadian Supplemental Type Certificate (STC) SH96–32 or United States STC SR00429 NY.

Note 1: This AD applies to each helicopter identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For helicopters that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required within 50 hours time-in-service or 30 days, whichever occurs first, unless accomplished previously.

To prevent inadvertent engagement of a collective control locking device, the collective pitch control locking in the fulldown position, and subsequent loss of control of the helicopter, accomplish the following:

- (a) Replace the collective control locking device with a redesigned locking device in accordance with the Accomplishment Instructions, paragraph 2, of ECL AS350 BA, B2 Service Bulletin No. ECL-99-67-002, Revision 2, dated September 23, 1999.
- (b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Regulations Group, Rotorcraft Directorate, FAA. Operators shall submit their requests through an FAA Principal Maintenance Inspector, who may concur or comment and then send it to the Manager, Regulations Group.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Regulations Group.

- (c) Special flight permits may be issued in accordance with 14 CFR 21.197 and 21.199 to operate the helicopter to a location where the requirements of this AD can be accomplished.
- (d) Replacing the collective locking device shall be done in accordance with the Accomplishment Instructions, paragraph 2, of Eurocopter Canada Limited Service Bulletin AS 350 BA, B2, No. ECL-99-67-002, Revision 2, dated September 23, 1999. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Eurocopter Canada Limited, 1100 Gilmore Rd., Fort Erie, Ontario L2A 5M-4005, telephone (972) 641-3460, fax (972) 641-3527. Copies may be inspected at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.
- (e) This amendment becomes effective on March 18, 2002.

Note 3: The subject of this AD is addressed in Transport Canada (Canada) AD CF-2000-06R1, dated August 23, 2000.

Issued in Fort Worth, Texas, on February 19, 2002.

Mark R. Schilling,

Acting Manager, Rotorcraft Directorate, Aircraft Certification Service.

[FR Doc. 02-4555 Filed 2-27-02; 8:45 am]

BILLING CODE 4910-13-U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001-NM-252-AD; Amendment 39-12667; AD 2002-04-10]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A319 Series Airplanes and A320-200 Series Airplanes

AGENCY: Federal Aviation Administration, DOT. **ACTION:** Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain Airbus Model A319 series airplanes and A320-200 series airplanes, that requires repetitive inspections to detect loose or missing rivets in specified areas of the door frames of the overwing emergency exits and corrective action, if necessary. This AD also requires measurement of the grip length of all rivets in the specified areas and corrective action, if necessary, which terminates the repetitive inspections. This amendment is prompted by mandatory continuing airworthiness information from a foreign airworthiness authority. The actions specified by this AD are intended to detect and correct loose or missing rivets or discrepant rivets, which could lead to reduced structural integrity of the overwing emergency exit door frames. This action is intended to address the identified unsafe condition.

DATES: Effective April 5, 2002.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of April 5, 2002.

ADDRESSES: The service information referenced in this AD may be obtained from Airbus Industrie, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. This information may be examined at the Federal Aviation Administration (FAA), Transport

Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Dan Rodina, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2125; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A

proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Model A319 series airplanes and A320–200 series airplanes was published in the Federal Register on November 23, 2001 (66 FR 58682). That action proposed to require repetitive inspections to detect loose or missing rivets in specified areas of the door frames of the overwing emergency exits and corrective action, if necessary. That action also proposed to require measurement of the grip length of all rivets in the specified areas and corrective action, if necessary, which would terminate the repetitive inspections.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Cost Impact

The FAA estimates that 168 Model A319 and A320–200 series airplanes of U.S. registry will be affected by this AD, that it will take approximately 1 work hour per airplane to accomplish the required inspection, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of this AD on U.S. operators is estimated to be \$10,080, or \$60 per airplane, per inspection cycle.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish