

“significant regulatory action” under Executive Order 12866; (2) is not a “significant rule” under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing amendment 39-9656 (61 FR 29465, June 11, 1996), and by adding a new airworthiness directive (AD), amendment 39-12580, to read as follows:

2001-26-14 Dornier Luftfahrt GMBH:

Amendment 39-12580. Docket 97-NM-187-AD. Supersedes AD 96-12-13, Amendment 39-9656.

Applicability: Model 328-100 airplanes, equipped with a Honeywell GP-300 guidance and display controller having part number (P/N) 7015327-901 or -902; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent a defective light bulb from causing a short circuit that emits smoke and fumes into the cockpit, or causing damage to the circuit cards and various components, which may lock the autopilot into the engaged mode, accomplish the following:

Restatement of Requirements of AD 96-12-13

(a) Within 60 days after June 26, 1996 (the effective date of AD 96-12-13, amendment 39-9656), modify the Honeywell GP-300 guidance and display controller, having P/N 7015327-901 or -902, in accordance with Honeywell Service Bulletin 7015327-22-2, dated March 4, 1996.

New Requirements of This AD

(b) Within 60 days after the effective date of this AD, verify that the wiring of the Honeywell GP-300 guidance and display controller is correct by conducting a re-test of the circuit card assemblies, in accordance with Honeywell Service Bulletin 7015327-22-4, dated March 31, 1997. If any discrepancy is found, prior to further flight, repair in accordance with the service bulletin.

Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

Special Flight Permits

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(e) The actions shall be done in accordance with Honeywell Service Bulletin 7015327-22-2, dated March 4, 1996; and Honeywell Service Bulletin 7015327-22-4, dated March 31, 1997.

(1) The incorporation by reference of Honeywell Service Bulletin 7015327-22-4, dated March 31, 1997, is approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51.

(2) The incorporation by reference of Honeywell Service Bulletin 7015327-22-2, dated March 4, 1996, was approved previously by the Director of the Federal Register as of June 26, 1996 (61 FR 29465, June 11, 1996).

(3) Copies may be obtained from Honeywell, Inc., Attn: Customer Support Materiel, PO Box 21111, Phoenix, Arizona 85036. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind

Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 3: The subject of this AD is addressed in German airworthiness directive 96-239/2, dated June 19, 1997.

Effective Date

(f) This amendment becomes effective on February 8, 2002.

Issued in Renton, Washington, on December 21, 2001.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 02-143 Filed 1-3-02; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NM-161-AD; Amendment 39-12581; AD 2001-26-15]

RIN 2120-AA64

Airworthiness Directives; McDonnell Douglas Model DC-9-81, -82, -83, and -87 Series Airplanes, and Model MD-88 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain McDonnell Douglas Model DC-9-81, -82, -83, and -87 series airplanes, and Model MD-88 airplanes, that requires a detailed visual inspection of certain wires to detect chafing and preload; repair, if necessary; and modification of certain wire assemblies. This action is necessary to prevent insufficient clearance between wire assemblies and the ice protection airduct and airstair door interlock rod; chafing; and consequent arcing of wire assemblies. Such arcing could result in damage to electronic equipment and adjacent structures, or cause the insulation blankets to ignite, which could result in smoke and fire in the flight deck and main cabin. This action is intended to address the identified unsafe condition.

DATES: Effective February 8, 2002.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of February 8, 2002.

ADDRESSES: The service information referenced in this AD may be obtained from Boeing Commercial Aircraft

Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Data and Service Management, Dept. C1-L5A (D800-0024). This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Elvin Wheeler, Aerospace Engineer, Systems and Equipment Branch, ANM-130L, FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712-4137; telephone (562) 627-5344; fax (562) 627-5210.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain McDonnell Douglas Model DC-9-81, -82, -83, and -87 series airplanes, and Model MD-88 airplanes was published in the **Federal Register** on June 5, 2001 (66 FR 30099). That action proposed to require a detailed visual inspection of certain wires to detect chafing and preload; repair, if necessary; and modification of certain wire assemblies.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comments received.

Request To Use a Previous Revision

One commenter request that the applicability of the proposed AD be revised to exclude airplanes on which the airstairs have been removed and the doors have been bolted shut. The commenter states that it is in compliance with McDonnell Douglas Service Bulletin MD80-24-126, Revision 01, dated May 14, 1998. The commenter notes McDonnell Douglas Alert Service Bulletin MD80-24A126, Revision 02, dated September 22, 1999 (which is referenced in the proposed AD as the appropriate source of service information), requires additional work, consisting of moving the airstair door to the open and closed position to ensure clearance of wiring. Therefore, the commenter concludes that the additional work in Revision 02 of the service bulletin is not necessary for its fleet.

The FAA does not agree with this commenter. A door bolted shut would clearly not allow inspection for wire clearance with the door in the open position. However, since the inspection required by this AD is also for clearance when the door is in the full-closed position, that part of the inspection would still be applicable. Also, additional wires need to be inspected per Revision 02 of McDonnell Douglas Alert Service Bulletin MD80-24A126. The fact that airplane structure has been altered does not necessarily mean that an unsafe condition has been eliminated or does not exist. Therefore, we find no change to the final rule is necessary in this regard. However, under the provisions of paragraph (b) of the final rule, we may consider requests for approval of an alternative method of compliance if sufficient data are submitted to substantiate that such a design change would provide an acceptable level of safety.

Request for Change of Level of Inspection

The same commenter also requests that the term "detailed visual inspection" be changed to "visual inspection," because the Common Support Data Dictionary (CSD) has a specific definition of "detailed visual inspection" and McDonnell Douglas Service Bulletin MD80-24A126, Revision 02, does not call for this level of inspection.

The FAA does not concur. The definition of the "detailed visual inspection" specified in **Note 2** of this AD is a standard definition that is used in all ADs that require a detailed visual inspection. We have determined that a detailed visual inspection is the appropriate level of inspection to be carried out for the unsafe condition being addressed.

Conclusion

After careful review of the available data, including the comments noted above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Cost Impact

There are approximately 1,037 Model DC-9-81, -82, -83, and -87 series airplanes, and Model MD-88 airplanes of the affected design in the worldwide fleet. The FAA estimates that 830 airplanes of U.S. registry will be affected by this AD, that it will take approximately 1 work hour per airplane to accomplish the required detailed visual inspection, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the

inspection required by this AD on U.S. operators is estimated to be \$49,800, or \$60 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

2001-26-15 McDonnell Douglas:

Amendment 39-12581. Docket 2000-NM-161-AD.

Applicability: Model DC-9-81, -82, -83, and -87 series airplanes, and Model MD-88 airplanes, as listed in McDonnell Douglas Alert Service Bulletin MD80-24A126, Revision 02, dated September 22, 1999; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent smoke and fire in the flight deck and main cabin due to insufficient clearance between wire assemblies and the ice protection airduct and airstair door interlock rod; chafing; and consequent arcing of wire assemblies, accomplish the following:

Inspection and Modification

(a) Within 6 months after the effective date of this AD, perform a detailed visual inspection of wire runs in the electrical/equipment compartment to detect chafing and preload against the airduct shroud assembly of the strake ice protection system and/or airstair door interlock rod between stations Y=148.00 and Y=160.000, in accordance with McDonnell Douglas Alert Service Bulletin MD80-24A126, Revision 02, dated September 22, 1999.

Note 2: For the purposes of this AD, a detailed visual inspection is defined as: "An intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required."

(1) If no chafed or preloaded wire is found, prior to further flight, install spacers, sta-straps, and tie-back wire bundles, in accordance with the service bulletin.

(2) If any chafed or preloaded wire is found, prior to further flight, repair, and install spacers, sta-straps, and tie-back wire bundles, in accordance with the service bulletin.

Alternative Methods of Compliance

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Los Angeles Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Los Angeles ACO.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Los Angeles ACO.

Special Flight Permit

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(d) The actions shall be done in accordance with McDonnell Douglas Alert Service Bulletin MD80-24A126, Revision 02, dated September 22, 1999. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Data and Service Management, Dept. C1-L5A (D800-0024). Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Effective Date

(e) This amendment becomes effective on February 8, 2002.

Issued in Renton, Washington, on December 21, 2001.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.
[FR Doc. 02-144 Filed 1-3-02; 8:45 am]

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DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. 2000-NM-162-AD; Amendment 39-12582; AD 2001-26-16]

RIN 2120-AA64

Airworthiness Directives; McDonnell Douglas DC-9-81, -82, -83, and -87 Series Airplanes, and Model MD-88 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain McDonnell Douglas DC-9-81, -82, -83, and -87 series airplanes, and Model MD-88 airplanes, that requires replacing the interface connectors of the cabin fluorescent lighting ballast in the wiring harness of the overhead stowage compartment with new connectors. In lieu of the required replacement, this AD requires adding interface seals to the existing interface connectors of the cabin fluorescent lighting ballast between certain stations and reidentifying the connector assemblies. This action is necessary to prevent electrical shorting and arcing due to the presence of water in the lighting ballast interface connectors, which could result in smoke in the main cabin. This action is intended to address the identified unsafe condition.

DATES: Effective February 8, 2002.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of February 8, 2002.

ADDRESSES: The service information referenced in this AD may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Data and Service Management, Dept. C1-L5A (D800-0024). This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Elvin Wheeler, Aerospace Engineer, Systems and Equipment Branch, ANM-