

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 23**

[Docket No. CE159; Notice No. 23-00-01-SC]

Special Conditions: Cessna Models; Diamond Model; Mooney Models; Piper Models; Raytheon Models; Airplanes Modified by Installation of Teledyne Continental Motors Full Authority Digital Engine Control (FADEC) System**AGENCY:** Federal Aviation Administration (FAA), DOT.**ACTION:** Notice of proposed special conditions.

SUMMARY: This notice proposes special conditions for the Cessna Models 172/K/L/M/N/P, 177/A/B/RG, 180/E/F/G/H/J/K, 182/E/F/G/H/J/K/L/M/N/P/Q/R, 185/A/C/D/E/F, 188/A/B/C, P206/A/B/C/D/E, U206/A/B/C/D/E/F/G, TU206/A/B/C/D/E/F/G, TP206/A/B/C/D/E, 207/A, T207/A, 210/K/L/M/N/R, T210/K/L/M/N/R, 310/A/B/C/D/E/F/G/H/I/J/J-1/K/L/N/P/Q/R, 320/A/B/C/D/E/F/-1, 337/A/B/C/D/E/F/G/H, 340/A, 401/A/B, 411/A, 414/A, 421/A/B/C; Diamond Model DA20-C1; Mooney Models M20/C/D/E/F/J/K/R; Piper Models PA-28-180/-201T, PA-28R-201T, PA-28RT-201T, PA-34-200/-200T/-220T, PA-46-310P/-350P; and Raytheon Models F33, V35, A36, 95-C55, D55, E55, 58, 58P airplanes as modified by Teledyne Continental Motors to include a FADEC System. These airplanes, as modified, will have a novel or unusual design feature associated with the installation of an engine that uses an electronic engine control system in place of the engine's mechanical system. The applicable airworthiness regulations do not contain adequate or appropriate safety standards for this design feature. These proposed special conditions contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

DATES: Comments must be received on or before May 4, 2000.**ADDRESSES:** Comments on this proposal may be mailed in duplicate to: Federal Aviation Administration, Regional Counsel, ACE-7, Attention: Rules Docket, Docket No. CE159, DOT Building, 901 Locust, Kansas City, Missouri 64106, or delivered in duplicate to the Regional Counsel at the above address. Comments must be marked: Docket No. CE159. Comments may be inspected in the Rules Docket

weekdays, except Federal holidays, between 7:30 a.m. and 4:00 p.m.

FOR FURTHER INFORMATION CONTACT:

Randy Griffith, Aerospace Engineer, Federal Aviation Administration, Aircraft Certification Service, Small Airplane Directorate, ACE-111, 901 Locust, Room 301, Kansas City, Missouri, 816-329-4126, fax 816-329-4090.

SUPPLEMENTARY INFORMATION:**Comments Invited**

Interested persons are invited to participate in the making of these proposed special conditions by submitting such written data, views, or arguments as they may desire. Communications should identify the regulatory docket or notice number and be submitted in duplicate to the address specified above. All communications received on or before the closing date for comments will be considered by the Administrator. If a comment applies to a specific airplane model, please identify the model in the comment. The proposals described in this notice may be changed in light of the comments received. All comments received will be available in the Rules Docket for examination by interested persons, both before and after the closing date for comments. A report summarizing each substantive public contact with FAA personnel concerning this rulemaking will be filed in the docket. Persons wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must include with those comments a self-addressed, stamped postcard on which the following statement is made: "Comments to CE159." The postcard will be date stamped and returned to the commenter.

Background

On January 7, 2000, Teledyne Continental Motors applied for supplemental type certificates for the installation of engines which use an electronic engine control system in place of the hydromechanical control system for the Cessna Models 172/K/L/M/N/P, 177/A/B/RG, 180/E/F/G/H/J/K, 182/E/F/G/H/J/K/L/M/N/P/Q/R, 185/A/C/D/E/F, 188/A/B/C, P206/A/B/C/D/E, U206/A/B/C/D/E/F/G, TU206/A/B/C/D/E/F/G, TP206/A/B/C/D/E, 207/A, T207/A, 210/K/L/M/N/R, T210/K/L/M/N/R, 310/A/B/C/D/E/F/G/H/I/J/J-1/K/L/N/P/Q/R, 320/A/B/C/D/E/F/-1, 337/A/B/C/D/E/F/G/H, 340/A, 401/A/B, 411/A, 414/A, 421/A/B/C; Diamond Model DA20-C1; Mooney Models M20/C/D/E/F/J/K/R; Piper Models PA-28-180/-201T, PA-28R-201T, PA-28RT-201T,

PA-34-200/-200T/-220T, PA-46-310P/-350P; and Raytheon Models F33, V35, A36, 95-C55, D55, E55, 58, 58P airplanes. Affected airplane models are currently approved under the following Type Certificate Numbers:

Model	Type Certificate No.
Cessna Models 172/K/L/M/N/P.	3A12
Cessna Models 177/A/B.	A13CE
Cessna Model 177RG.	A20CE
Cessna Models 180/E/F/G/H/J/K.	5A6
Cessna Models 182/E/F/G/H/J/K/L/M/N/P/Q/R.	3A13
Cessna Models 185/A/C/D/E/F.	3A24
Cessna Models 188/A/B/C.	A9CE
Cessna Models P206/A/B/C/D/E, U206/A/B/C/D/E/F/G, TU206/A/B/C/D/E/F/G, TP206/A/B/C/D/E.	A4CE
Cessna Models 207/A, T207/A.	A16CE
Cessna Models 210/K/L/M/N/R, T210/K/L/M/N/R.	3A21
Cessna Model 310/A/B/C/D/E/F/G/H/I/J/J-1/K/L/N/P/Q/R.	3A10
Cessna Models 320/A/B/C/D/E/F/-1, 340/A.	3A25
Cessna Model 337/A/B/C/D/E/F/G/H.	A6CE
Cessna Models 401/A/B, 411/A, 414/A, 421/A/B/C.	A7CE
Diamond Model DA20-C1.	TA4CH
Mooney Models M20/C/D/E/F/J/K/R.	2A3
Piper Models PA-28-180/-201T, PA-28R-201T, PA-28RT-201T.	2A13
Piper Model PA-34-200/-200T/-220T.	A7SO
Piper Model PA-46-310P/-350P.	A25SO
Raytheon Models F33, V35, A36.	3A15
Raytheon Models 58, 95-C55, D55, E55.	3A16
Raytheon Model 58P	A23CE

All the airplanes are small, normal category airplanes powered with either single or dual reciprocating engines. The modification to the airplanes involves replacement of the engine with a new engine model that incorporates an electronic engine control system with full engine authority capability. The new engine model is accomplished with either an amended type certificate to the

engine if the engine is a Teledyne Continental engine or a supplemental type certificate to the engine if the engine is a Lycoming engine. The airframe systems will also be modified as necessary to accommodate the engine's new control system.

Type Certification Basis

Under the provisions of § 21.101, Teledyne Continental Motors must show that affected airplane models, as changed, continue to meet the applicable provisions of the regulations incorporated by reference in Type Certificate Numbers 3A12, A13CE, A20CE, 5A6, 3A13, 3A24, A9CE, A4CE, A16CE, 3A21, 3A10, 3A25, A6CE, A7CE, TA4CH, 2A3, 2A13, A7SO, A25SO, 3A15, 3A16, A23CE or the applicable regulations in effect on the date of application for the change. The regulations incorporated by reference in the type certificate are commonly referred to as the "original type certification basis" and can be found in the following Type Certificate Numbers:

Model	Type Certificate Number
Cessna Models 172/K/L/M/N/P.	3A12 Rev 65; Dec 15, 99
Cessna Models 177/A/B	A13CE Rev 23; Oct 15, 94
Cessna Model 177RG	A20CE Rev 18; Oct 15, 94
Cessna Models 180/E/F/G/H/J/K.	5A6 Rev 62; Jun 15, 95
Cessna Models 182/E/F/G/H/J/K/L/M/N/P/Q/R.	3A13 Rev 56; Dec 15, 99
Cessna Models 185/A/C/D/E/F.	3A24 Rev 36; Nov 15, 99
Cessna Models 188/A/B/C	A9CE Rev 26; Oct 15, 95
Cessna Models P206/A/B/C/D/E, U206/A/B/C/D/E/F/G, TU206/A/B/C/D/E/F/G, TP206/A/B/C/D/E.	A4CE Rev 37; Dec 15, 94
Cessna Models 207/A, T207/A.	A16CE Rev 20; Oct 15, 94
Cessna Models 210/K/L/M/N/R, T210/K/L/M/N/R.	3A21 Rev 45; Aug 15, 96
Cessna Model 310/A/B/C/D/E/F/G/H/I/J/J-1/K/L/N/P/Q/R.	3A10 Rev 61; Nov 15, 97
Cessna Models 320/A/B/C/D/E/F/-1, 340/A.	3A25 Rev 25; Aug 15, 94
Cessna Model 337/A/B/C/D/E/F/G/H.	A6CE Rev 37; Oct 15, 94
Cessna Models 401/A/B, 411/A, 414/A, 421/A/B/C.	A7CE Rev 44; May 15, 99
Diamond Model DA20-C1	TA4CH Rev 4; Apr 8, 99
Mooney Models M20/C/D/E/F/J/K/R.	2A3 Rev 46; Aug 10, 99
Piper Models PA-28-180/-201T, PA-28R-201T, PA-28RT-201T.	2A13 Rev 44; Oct 15, 97

Model	Type Certificate Number
Piper Model PA-34-200/-200T/-220T.	A7SO Rev 13; Dec 18, 96
Piper Model PA-46-310P/-350P.	A25SO Rev 8; Mar 4, 99
Raytheon Models F33, V35, A36.	3A15 Rev 88; Jan 15, 00
Raytheon Models 58, 95-C55, D55, E55.	3A16 Rev 80; Jan 18, 00
Raytheon Model 58P	A23CE Rev 14; Apr 15, 96

If the Administrator finds that the applicable airworthiness regulations (i.e., Part 23) do not contain adequate or appropriate safety standards for affected airplane models because of a novel or unusual design feature, special conditions are prescribed under the provisions of § 21.16.

Special conditions, as appropriate, are issued in accordance with § 11.49 after public notice, as required by §§ 11.28 and 11.29(b), and become part of the type certification basis in accordance with § 21.101(b)(2).

Special conditions are initially applicable to the model for which they are issued. Should the applicant apply for a supplemental type certificate to modify any other model included on the same type certificate to incorporate the same novel or unusual design feature, the special conditions would also apply to the other model under the provisions of § 21.101(a)(1).

Novel or Unusual Design Features

The Cessna Models 172/K/L/M/N/P, 177/A/B/RG, 180/E/F/G/H/J/K, 182/E/F/G/H/J/K/L/M/N/P/Q/R, 185/A/C/D/E/F, 188/A/B/C, P206/A/B/C/D/E, U206/A/B/C/D/E/F/G, TU206/A/B/C/D/E/F/G, TP206/A/B/C/D/E, 207/A, T207/A, 210/K/L/M/N/R, T210/K/L/M/N/R, 310/A/B/C/D/E/F/G/H/I/J/J-1/K/L/N/P/Q/R, 320/A/B/C/D/E/F/-1, 337/A/B/C/D/E/F/G/H, 340/A, 401/A/B, 411/A, 414/A, 421/A/B/C; Diamond Model DA20-C1; Mooney Models M20/C/D/E/F/J/K/R; Piper Models PA-28-180/-201T, PA-28R-201T, PA-28RT-201T, PA-34-200/-200T/-220T, PA-46-310P/-350P; and Raytheon Models F33, V35, A36, 95-C55, D55, E55, 58, 58P airplanes will incorporate an engine that includes an electronic control system with full engine authority capability. The airframe systems will also be modified as necessary to accommodate the engine's new control system.

Many advanced electronic systems are prone to either upsets or damage, or both, at energy levels lower than analog systems. The increasing use of high power radio frequency emitters mandates requirements for improved

high intensity radiated fields (HIRF) protection for electrical and electronic equipment. Since the electronic engine control system developed by Teledyne Continental Motors will perform functions in which a failure may cause an unsafe condition, provisions for protection from the effects of HIRF fields should be considered and, if necessary, incorporated into the airplane design data. The FAA policy contained in Notice 8110.71, dated April 2, 1998, establishes the HIRF energy levels that airplanes will be exposed to in service. The guidelines set forth in this Notice are the result of an Aircraft Certification Service review of existing policy on HIRF, in light of the ongoing work of the ARAC Electromagnetic Effects Harmonization Working Group (EEHWG). The EEHWG adopted a set of HIRF environment levels in November 1997 that were agreed upon by the FAA, JAA, and industry participants. As a result, the HIRF environments in this notice reflect the environment levels recommended by this working group. This notice states that a full authority digital engine control is an example of a system that should address the HIRF environments.

Even though each control system will be certificated as part of the engine, the installation of an engine with an electronic control system requires evaluation due to the possible effects on or by other airplane systems (e.g., radio interference with other airplane electronic systems, shared engine and airplane power sources). The regulatory requirements in 14 CFR Part 23 for evaluating the installation of complex systems, including electronic systems, are contained in § 23.1309. However, when § 23.1309 was developed, the use of electronic control systems for engines was not envisioned; therefore, the § 23.1309 requirements were not applicable to systems certificated as part of the engine (reference § 23.1309(f)(1)). Also, electronic control systems often require inputs from airplane data and power sources and outputs to other airplane systems (e.g., automated cockpit powerplant controls such as mixture setting). Although the parts of the system that are not certificated with the engine could be evaluated using the criteria of § 23.1309, the integral nature of systems such as these makes it unfeasible to evaluate the airplane portion of the system without including the engine portion of the system. However, § 23.1309(f)(1) again prevents complete evaluation of the installed airplane system since evaluation of the engine system's effects is not required.

Therefore, special conditions are proposed for the Cessna Models 172/K/

L/M/N/P, 177/A/B/RG, 180/E/F/G/H/J/K, 182/E/F/G/H/J/K/L/M/N/P/Q/R, 185/A/C/D/E/F, 188/A/B/C, P206/A/B/C/D/E, U206/A/B/C/D/E/F/G, TU206/A/B/C/D/E/F/G, TP206/A/B/C/D/E, 207/A, T207/A, 210/K/L/M/N/R, T210/K/L/M/N/R, 310/A/B/C/D/E/F/G/H/I/J/J-1/K/L/N/P/Q/R, 320/A/B/C/D/E/F/-1, 337/A/B/C/D/E/F/G/H, 340/A, 401/A/B, 411/A, 414/A, 421/A/B/C; Diamond Model DA20-C1; Mooney Models M20/C/D/E/F/J/K/R; Piper Models PA-28-180/-201T, PA-28R-201T, PA-28RT-201T, PA-34-200/-200T/-220T, PA-46-310P/-350P; and Raytheon Models F33, V35, A36, 95-C55, D55, E55, 58, 58P airplanes modified by Teledyne Continental Motors by installation of an electronic engine control system to provide HIRF protection and to evaluate the installation of the electronic engine control system for compliance with the requirements of § 23.1309(a) through (e) at Amendment 23-41.

Applicability

As discussed above, these special conditions are applicable to the Cessna Models 172/K/L/M/N/P, 177/A/B/RG, 180/E/F/G/H/J/K, 182/E/F/G/H/J/K/L/M/N/P/Q/R, 185/A/C/D/E/F, 188/A/B/C, P206/A/B/C/D/E, U206/A/B/C/D/E/F/G, TU206/A/B/C/D/E/F/G, TP206/A/B/C/D/E, 207/A, T207/A, 210/K/L/M/N/R, T210/K/L/M/N/R, 310/A/B/C/D/E/F/G/H/I/J/J-1/K/L/N/P/Q/R, 320/A/B/C/D/E/F/-1, 337/A/B/C/D/E/F/G/H, 340/A, 401/A/B, 411/A, 414/A, 421/A/B/C; Diamond Model DA20-C1; Mooney Models M20/C/D/E/F/J/K/R; Piper Models PA-28-180/-201T, PA-28R-201T, PA-28RT-201T, PA-34-200/-200T/-220T, PA-46-310P/-350P; and Raytheon Models F33, V35, A36, 95-C55, D55, E55, 58, 58P airplanes as modified by Teledyne Continental Motors. Should Teledyne Continental Motors apply at a later date for a supplemental type certificate to modify any other model included on Type Certificate Numbers 3A12, A13CE, A20CE, 5A6, 3A13, 3A24, A9CE, A4CE, A16CE, 3A21, 3A10, 3A25, A6CE, A7CE, TA4CH, 2A3, 2A13, A7SO, A25SO, 3A15, 3A16, A23CE to incorporate the same novel or unusual design feature, the special conditions would apply to that model as well under the provisions of § 21.101(a)(1).

Conclusion

This action affects only certain novel or unusual design features on Cessna Models 172/K/L/M/N/P, 177/A/B/RG, 180/E/F/G/H/J/K, 182/E/F/G/H/J/K/L/M/N/P/Q/R, 185/A/C/D/E/F, 188/A/B/C, P206/A/B/C/D/E, U206/A/B/C/D/E/F/G, TU206/A/B/C/D/E/F/G, TP206/A/B/C/D/E, 207/A, T207/A, 210/K/L/M/N/R,

T210/K/L/M/N/R, 310/A/B/C/D/E/F/G/H/I/J/J-1/K/L/N/P/Q/R, 320/A/B/C/D/E/F/-1, 337/A/B/C/D/E/F/G/H, 340/A, 401/A/B, 411/A, 414/A, 421/A/B/C; Diamond Model DA20-C1; Mooney Models M20/C/D/E/F/J/K/R; Piper Models PA-28-180/-201T, PA-28R-201T, PA-28RT-201T, PA-34-200/-200T/-220T, PA-46-310P/-350P; and Raytheon Models F33, V35, A36, 95-C55, D55, E55, 58, 58P airplanes. It is not a rule of general applicability. It is only applicable to airplanes being modified by Teledyne Continental Motors to include this engine system.

List of Subjects in 14 CFR Part 23

Aircraft, Aviation safety, Signs and symbols.

Citation

The authority citation for these special conditions is as follows:

Authority: 49 U.S.C. 106(g), 40113 and 44701; 14 CFR 21.16 and 21.101 and 14 CFR 11.28 and 11.29(b).

The Proposed Special Conditions

Accordingly, the Federal Aviation Administration (FAA) proposes the following special conditions as part of the type certification basis for Cessna Models 172/K/L/M/N/P, 177/A/B/RG, 180/E/F/G/H/J/K, 182/E/F/G/H/J/K/L/M/N/P/Q/R, 185/A/C/D/E/F, 188/A/B/C, P206/A/B/C/D/E, U206/A/B/C/D/E/F/G, TU206/A/B/C/D/E/F/G, TP206/A/B/C/D/E, 207/A, T207/A, 210/K/L/M/N/R, T210/K/L/M/N/R, 310/A/B/C/D/E/F/G/H/I/J/J-1/K/L/N/P/Q/R, 320/A/B/C/D/E/F/-1, 337/A/B/C/D/E/F/G/H, 340/A, 401/A/B, 411/A, 414/A, 421/A/B/C; Diamond Model DA20-C1; Mooney Models M20/C/D/E/F/J/K/R; Piper Models PA-28-180/-201T, PA-28R-201T, PA-28RT-201T, PA-34-200/-200T/-220T, PA-46-310P/-350P; and Raytheon Models F33, V35, A36, 95-C55, D55, E55, 58, 58P airplanes modified by Teledyne Continental Motors to include an engine with a FADEC System.

1. *High Intensity Radiated Fields (HIRF) Protection.* In showing compliance with 14 CFR Part 21 and the airworthiness requirements of 14 CFR Part 23, protection against hazards caused by exposure to HIRF fields for the full authority digital engine control system, which performs functions in which a failure may cause an unsafe condition to the airplane, must be considered. To prevent this occurrence, the electronic engine control system, must be designed and installed to ensure that the operation and operational capabilities of this critical system isare not adversely affected

when the airplane is exposed to high energy radio fields.

At this time, the FAA and other airworthiness authorities are unable to precisely define or control the HIRF energy level to which the airplane will be exposed in service; therefore, the FAA hereby defines two acceptable interim methods for complying with the requirement for protection of systems that perform functions in which a failure may cause an unsafe condition.

(1) The applicant may demonstrate that the operation and operational capability of the installed electrical and electronic systems that perform functions in which a failure may cause an unsafe condition, are not adversely affected when the aircraft is exposed to the external HIRF threat environment defined in the following table:

Frequency	Field Strength (volts per meter)	
	Peak	Average
10 kHz–100 kHz	50	50
100 kHz–500 kHz	50	50
500 kHz–2 MHz	50	50
2 MHz–30 MHz	100	100
30 MHz–70 MHz	50	50
70 MHz–100 MHz	50	50
100 MHz–200 MHz	100	100
200 MHz–400 MHz	100	100
400 MHz–700 MHz	700	50
700 MHz–1 GHz	700	100
1 GHz–2 GHz	2000	200
2 GHz–4 GHz	3000	200
4 GHz–6 GHz	3000	200
6 GHz–8 GHz	1000	200
8 GHz–12 GHz	3000	300
12 GHz–18 GHz	2000	200
18 GHz–40 GHz	600	200

The field strengths are expressed in terms of peak root-mean-square (rms) values.

or,

(2) The applicant may demonstrate by a system test and analysis that the electrical and electronic systems that perform functions in which a failure may cause an unsafe condition can withstand a minimum threat of 100 volts per meter peak electrical strength, without the benefit of airplane structural shielding, in the frequency range of 10 KHz to 18 GHz.

When using this test to show compliance with the HIRF requirements, no credit is given for signal attenuation due to installation.

2. *Electronic Engine Control System.* The installation items that affect the electronic engine control system must comply with the requirements of

§ 23.1309 (a) through (e) at Amendment 23-41.

Issued in Kansas City, Missouri on March 22, 2000.

Michael Gallagher,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 00-8231 Filed 4-3-00; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Airspace Docket No. 99-ANM-12]

Proposed Revision of Class E Airspace, North Bend, OR

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of Proposed Rulemaking (NPRM).

SUMMARY: This proposal would amend the North Bend, OR, Class E airspace to accommodate the development of a revised Standard Instrument Approach Procedure (SIAP) at the North Bend Municipal Airport, North Bend, OR.

DATES: Comments must be received on or before May 19, 2000.

ADDRESSES: Send comments on the proposal in triplicate to: Manager, Airspace Branch, ANM-520, Federal Aviation Administration, Docket No. 99-ANM-12, 1601 Lind Avenue SW, Renton, Washington 98055-4056.

The official docket may be examined in the office of the Assistant Chief Counsel for the Northwest Mountain Region at the same address.

An informal docket may also be examined during normal business hours in the office of the Manager, Air Traffic Division, Airspace Branch, at the address listed above.

FOR FURTHER INFORMATION CONTACT: Dennis Ripley, ANM-520.6, Federal Aviation Administration, Docket No. 99-ANM-12, 1601 Lind Avenue SW, Renton, Washington 98055-4056; telephone number: (425) 227-2527.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested parties are invited to participate in this proposed rulemaking by submitting such written data, views, or arguments, as they may desire. Comments that provide the factual basis supporting the views and suggestions presented are particularly helpful in developing reasoned regulatory decisions on the proposal. Comments are specifically invited on the overall

regulatory, aeronautical, economic, environmental, and energy related aspects of the proposal.

Communications should identify the airspace docket number and be submitted in triplicate to the address listed above. Commenters wishing the FAA to acknowledge receipt of their comments on this notice must submit, with those comments, a self-addressed stamped postcard on which the following statement is made: "Comments to Airspace Docket No. 99-ANM-12." The postcard will be date/time stamped and returned to the commenter. All communications received on or before the specified closing date for comments will be considered before taking action on the proposed rule. The proposal contained in this notice may be changed in the light of comments received. All comments submitted will be available for examination at the address listed above both before and after the closing date for comments. A report summarizing each substantive public contact with FAA personnel concerned with this rulemaking will be filed in the docket.

Availability of NPRM's

Any person may obtain a copy of this NPRM by submitting a request to the Federal Aviation Administration, Airspace Branch, ANM-520, 1601 Lind Avenue SW, Renton, Washington 98055-4056. Communications must identify the notice number of this NPRM. Persons interested in being placed on a mailing list for future NPRM's should also request a copy of Advisory Circular No. 11-2A, which describes the application procedure.

The Proposal

The FAA is considering an amendment to Title 14 Code of Federal Regulations, part 71 (14 CFR part 71) by revising Class E airspace at North Bend, OR, in order to accommodate a revised SIAP to the North Bend Municipal Airport, North Bend, OR. This amendment would provide additional airspace at North Bend, OR, to meet current criteria standards associated with SIAP holding patterns. The FAA establishes Class E airspace where necessary to contain aircraft transitioning between the terminal and en route environments. The intended effect of this proposal is designed to provide for the safe and efficient use of the navigable airspace. This proposal would promote safe flight operations under Instrument Flight Rules (IFR) at the North Bend Airport and between the terminal and en route transition stages.

The area would be depicted on aeronautical charts for pilot reference. The coordinates for this airspace docket are based on North American Datum 83. Class E airspace areas extending upward from 700 feet or more above the surface of the earth, are published Paragraph 6005, of FAA Order 7400.9G dated September 1, 1999, and effective September 16, 1999, which is incorporated by reference in 14 CFR 71.1. The Class E airspace designation listed in this document would be published subsequently in the Order.

The FAA has determined that this proposed regulation only involves an established body of technical regulations for which frequent and routine amendments are necessary to keep them operationally current. It, therefore, (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034; February 26, 1979); and (3) does not warrant preparation of a Regulatory Evaluation as the anticipated impact is so minimal. Since this is a routine matter that will only affect air traffic procedures and air navigation, it is certified that this rule, when promulgated, will not have a significant economic impact on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 71

Airspace, Incorporation by reference, Navigation (air).

The Proposed Amendment

In consideration of the foregoing, the Federal Aviation Administration proposes to amend 14 CFR part 71 as follows:

PART 71—DESIGNATION OF CLASS A, CLASS B, CLASS C, CLASS D, AND CLASS E AIRSPACE AREAS; AIRWAYS; ROUTES; AND REPORTING POINTS

1. The authority citation for 14 CFR part 71 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40103, 40113, 40120; E.O. 10854, 24 FR 9565, 3 CFR, 1959-1963 Comp., p. 389.

§ 71.1 [AMENDED]

2. The incorporation by reference in 14 CFR 71.1 of the Federal Aviation Administration Order 7400.9G, Airspace Designations and Reporting Points, dated September 1, 1999, and effective September 16, 1999, is amended as follows: