repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated in the body of this AD, unless already accomplished.

To prevent carbon monoxide gas from entering the airplane's cabin heating system and cabin, which, if not corrected, could result in passenger and pilot injury with consequent loss of control of the airplane, accomplish the following:

(a) Prior to further flight after the effective date of this AD, de-activate the cabin heating system by ensuring that the valve mechanism is functional, and that the cabin heat valve lever is safety wired in the down "off" position.

(b) Prior to further flight after the effective date of this AD, fabricate and install a placard near the cabin heat control knob, within the pilot's clear view, using at least \(^1/8\)-inch letters with the following words: "CABIN HEATER INOPERATIVE"

(c) Within the next 50 hours time-inservice (TIS) after the effective date of this AD, replace the engine exhaust muffler with a muffler having one of the following part numbers (P/N) in accordance with the appropriate Cessna maintenance manual: 00624–NH4000011–10 71379 0554011–2–

PTT, or

0554011–2, or 0554011–6. or

an FAA-approved equivalent part number.

Note 3: P/N 0554011–2 will have "PT" stamped on the right-hand external ring that supports the muffler; and, P/N 0554011–6 may have "PT" stamped on the right-hand external ring.

- (d) If parts are not available for the replacement required in paragraph (c) of this AD, the airplane may continue to be operated for a period not to exceed 6 calendar months from the effective date of this AD, provided the cabin heating system remains deactivated.
- (e) The cabin heating system may be reactivated and the placard required in paragraph (b) of this AD may be removed, once the muffler is replaced in accordance with this AD.
- (f) Upon the effective date of this AD, no person may install any Aeroquip engine exhaust muffler, P/N 00624–NH4000011–10 71379 0554011–2, on any Cessna Model 172R airplane.
- (g) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished, provided the airplane cabin heater system is not used during that flight.
- (h) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Wichita Aircraft Certification Office, 1801 Airport Road, Rm. 100, Mid-Continent Airport, Wichita, Kansas, 67209. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then

send it to the Manager, Wichita Aircraft Certification Office.

Note 4: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Wichita Aircraft Certification Office.

- (i) Copies of this document may be inspected at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri.
- (j) This amendment (39–10306) becomes effective on February 23, 1998, to all persons except those persons to whom it was made immediately effective by priority letter AD 98–02–05, issued January 9, 1998, which contained the requirements of this amendment.

Issued in Kansas City, Missouri, on January 28, 1998.

Terry L. Chasteen,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 98–2774 Filed 2–4–98; 8:45 am] BILLING CODE 4910–13–U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 97-CE-85-AD; Amendment 39-10307; AD 98-03-14]

RIN 2120-AA64

Airworthiness Directives; EXTRA Flugzeugbau GmbH Models EA-300 and EA-300/S Airplanes

AGENCY: Federal Aviation Administration, DOT. **ACTION:** Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to EXTRA Flugzeugbau GmbH Models EA-300 and EA-300/S airplanes. This AD requires inspecting the upper longeron cutout bridge for cracks, repairing any cracks found, and modifying this area. This AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Germany. The actions specified by this AD are intended to prevent structural damage to the fuselage caused by cracks in the upper longeron cutout bridge, which, if not detected and corrected, could result in loss of control of the airplane.

DATES: Effective March 16, 1998. The incorporation by reference

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of March 16, 1998.

ADDRESSES: Service information that applies to this AD may be obtained from

EXTRA Flugzeugbau GmbH, Flugplatz Dinslaken, 46569 Hünxe, Germany. This information may also be examined at the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 97–CE–85–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Mr. Karl Schletzbaum, Aerospace Engineer, FAA, Small Airplane Directorate, 1201 Walnut Street, suite 900, Kansas City, Missouri 64106; telephone: (816) 426–6934; facsimile: (816) 426–2169.

SUPPLEMENTARY INFORMATION:

Events Leading to the Issuance of This AD

A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to EXTRA Flugzeugbau GmbH Models EA-300 and EA-300/S airplanes was published in the Federal Register as a notice of proposed rulemaking (NPRM) on November 5, 1997 (62 FR 59826). The NPRM proposed to require inspecting the upper longeron cutout bridge for cracks, repairing any cracks found, and modifying this area. Accomplishment of the proposed actions as specified in the NPRM would be required in accordance with EXTRA Service Bulletin No. 300-3-93, dated January 12, 1994.

The NPRM was the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Germany.

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were received on the proposed rule or the FAA's determination of the cost to the public.

The FAA's Determination

After careful review of all available information related to the subject presented above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed except for minor editorial corrections. The FAA has determined that these minor corrections will not change the meaning of the AD and will not add any additional burden upon the public than was already proposed.

Cost Impact

The FAA estimates that 68 airplanes in the U.S. registry will be affected by this AD, that it will take approximately 13 workhours (Inspection: 3 workhours; Modification: 10 workhours) per

airplane to accomplish the required action, and that the average labor rate is approximately \$60 an hour. Parts cost approximately \$200 per airplane. Based on these figures, the total cost impact of this AD on U.S. operators is estimated to be \$66,640, or \$980 per airplane.

Regulatory Impact

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a 'significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the final evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive (AD) to read as follows:

98-03-14 Extra Flugzeugbau GMBH:Amendment 39-10307; Docket No. 97-CE-85-AD.

Applicability: The following models and serial number airplanes, certificated in any category:

Model	Serial numbers
EA-300	V1 and 01 through 50.
EA-300/S	01 through 17.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated in the body of this AD, unless already accomplished.

To prevent structural damage to the fuselage caused by cracks in the upper longeron cutout bridge, which, if not detected and corrected, could result in loss of control of the airplane, accomplish the following:

(a) Upon accumulating 1,000 hours time-in-service (TIS) on the upper longeron or within the next 100 hours TIS after the effective date of this AD, whichever occurs later, inspect the upper longeron cutout bridge for cracks in accordance with the *Instructions* section of EXTRA Service Bulletin No. 300–3–93, dated January 12, 1994

(b) Prior to further flight after the inspection required by paragraph (a) of this AD, accomplish the following in accordance with the *Instructions* section of EXTRA Service Bulletin No. 300–3–93, dated January 12, 1994:

(1) Repair any cracks found in the upper longeron cut-out bridge; and

(2) Modify the upper longeron cut-out bridge.

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(d) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Small Airplane Directorate, FAA, 1201 Walnut, suite 900, Kansas City, Missouri 64106. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Small Airplane Directorate.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Small Airplane Directorate.

(e) Questions or technical information related to EXTRA Service Bulletin No. 300– 3–93, dated January 12, 1994, should be directed to EXTRA Flugzeugbau GmbH, Flugplatz Dinslaken, 46569 Hünxe, Germany. This service information may be examined at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri.

(f) The inspections, repairs, and modifications required by this AD shall be done in accordance with EXTRA Service Bulletin No. 300-3-93, dated January 12, 1994. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from EXTRA Flugzeugbau GmbH, Flugplatz Dinslaken, 46569 Hünxe, Germany. Copies may be inspected at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri, or at the Office of the Federal Register, 800 North Capitol Street, NW, suite 700, Washington, DC.

Note 3: The subject of this AD is addressed in German AD No. 94–043, dated October 21, 1994.

(g) This amendment (39–10307) becomes effective on March 16, 1998.

Issued in Kansas City, Missouri, on January $28,\,1998.$

Terry L. Chasteen,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 98–2775 Filed 2–4–98; 8:45 am] BILLING CODE 4910–13–U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 95

[Docket No. 29123; Amdt. No. 407]

IFR Altitudes; Miscellaneous Amendments

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts miscellaneous amendments to the required IFR (instrument flight rules) altitudes and changeover points for certain Federal airways, jet routes, or direct routes for which a minimum or maximum en route authorized IFR altitude is prescribed. This regulatory action is needed because of changes occurring in the National Airspace System. These changes are designed to provide for the safe and efficient use of the navigable airspace under instrument conditions in the affected areas.

EFFECTIVE DATE: 0901 UTC, February 26, 1998.

FOR FURTHER INFORMATION CONTACT: Paul J. Best, Flight Procedures Standards Branch (AFS–420), Technical Programs Division, Flight Standards Service Federal Aviation Administration, 800 Independence Avenue, SW.,